

# **Elective in Robotics**

coordinator: Prof. Giuseppe Oriolo

## **Visual Hovering of a quadrotor aerial vehicle**

(slides prepared by L. Rosa)

DIPARTIMENTO DI INFORMATICA  
E SISTEMISTICA ANTONIO RUBERTI



**SAPIENZA**  
UNIVERSITÀ DI ROMA

# Task

Visual hovering: regulate the position of a quadrotor vehicle (in hovering conditions) above a selected target, using visual feedback



Motivations:

GPS is unusable in indoor/cluttered environments

Image analysis can provide a lot of information

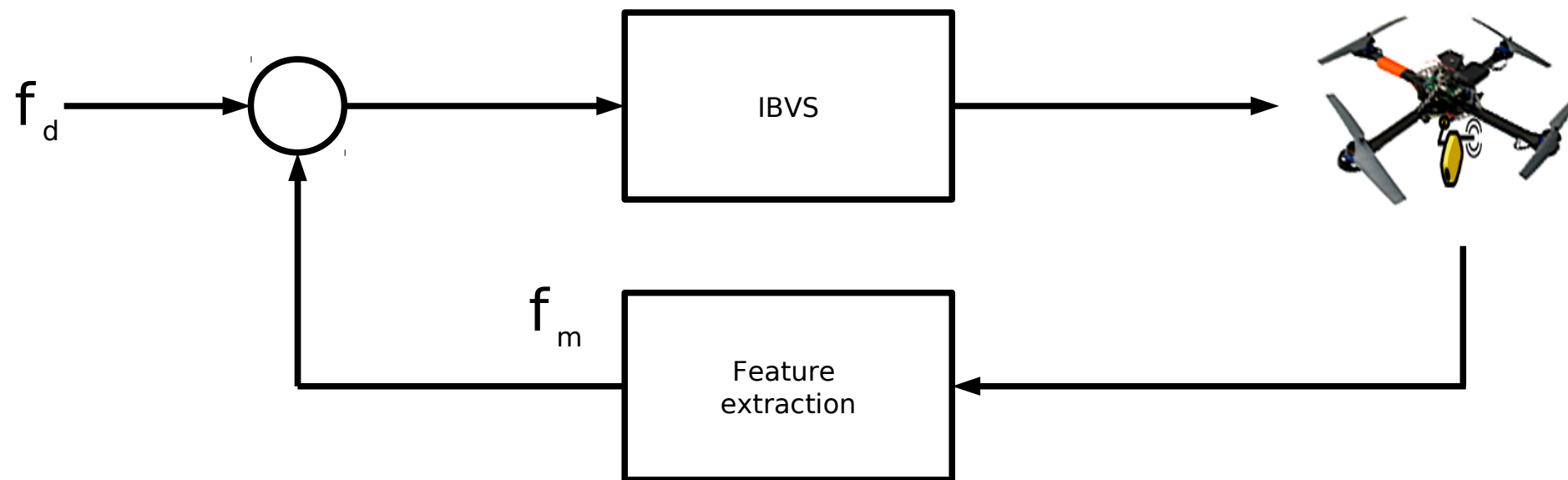
Note:

- Barometer is unreliable indoor
- Image analysis requires computational power

# Control scheme

Image Base Visual Servoing:

- Monocular camera (eye in hand configuration)
- IMU data (attitude/rotational velocities)



Visual error on image plane (tracking a desired target/object)



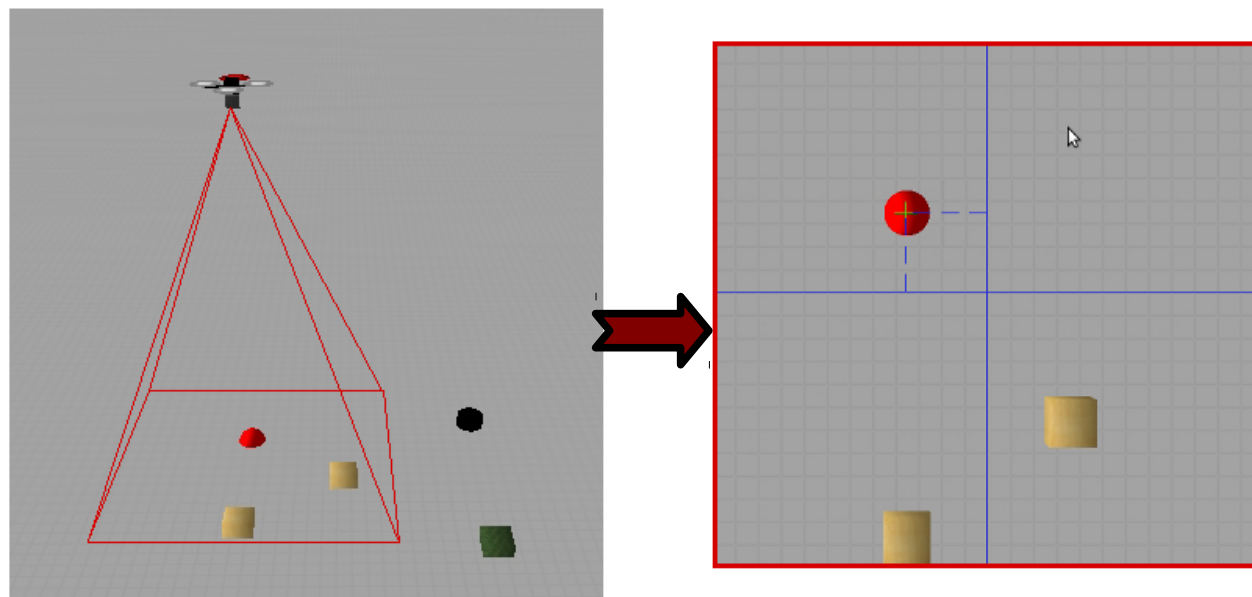
Do not need pose (and linear velocity) estimation

Reduced computation time

# Feature exctraction and tracking

Two algoritms tested:

- Camshift: based on analysis of color probability distribution
  - Robust to occlusion
  - Suffers light intensity variations (also shadows)
- VISP (IRISA project): based on analysis of image moments
  - Tracks elliptical targets
  - Robust to occlusion
  - Robust to light intensity variations



We get the error (on the image plane) and its derivative

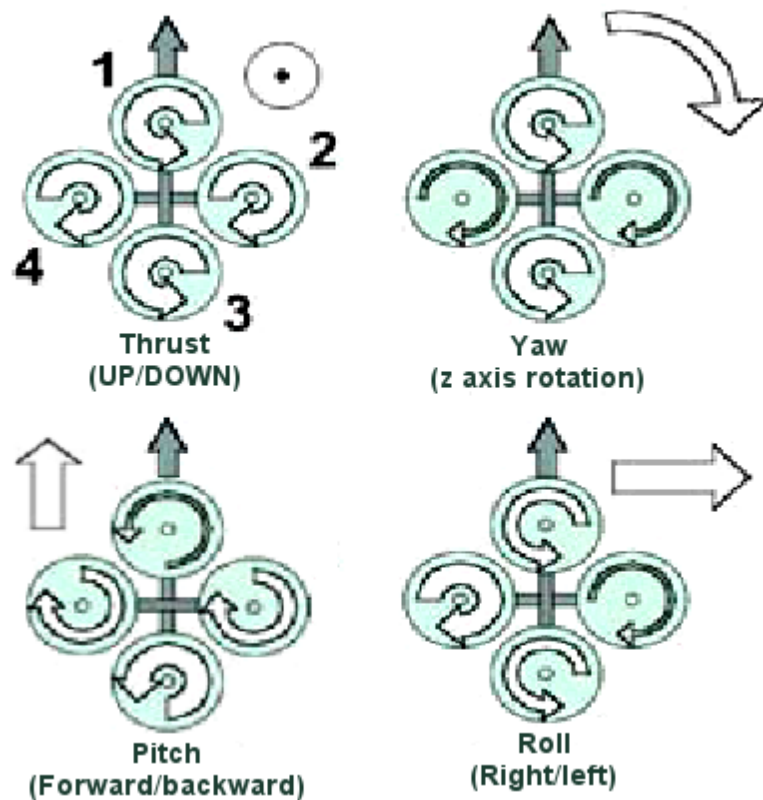
Note:

- The loss of the target leads to an undefined/wrong value
- Usually the reference is the center of the image (far from boundaries)

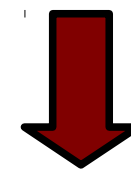
# Quadrotor motion

The vehicle achieves motion by varying its attitude

Symmetric motion



Hypothesis 1: a “fast” inner control loop is performing attitude control



Desired attitude is realized as soon as the vehicle receives control data.

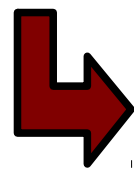
# Dynamic model

To develop control law, we can use the following simplified model

$$\begin{cases} \ddot{x} = \dot{V}_x = (s_\theta c_\phi) \frac{T}{m} \\ \ddot{y} = \dot{V}_y = -s_\phi \frac{T}{m} \\ \ddot{z} = \dot{V}_z = g - (c_\theta c_\phi) \frac{T}{m} \end{cases}$$

Hypotheses:

- Aerodynamic disturbances are neglected
- Inertial matrix is diagonal
- Camera center coincides with quadrotor center of gravity
- Yaw angle is zero
- Attitude controller is faster than visual control (1Khz vs 25Hz)

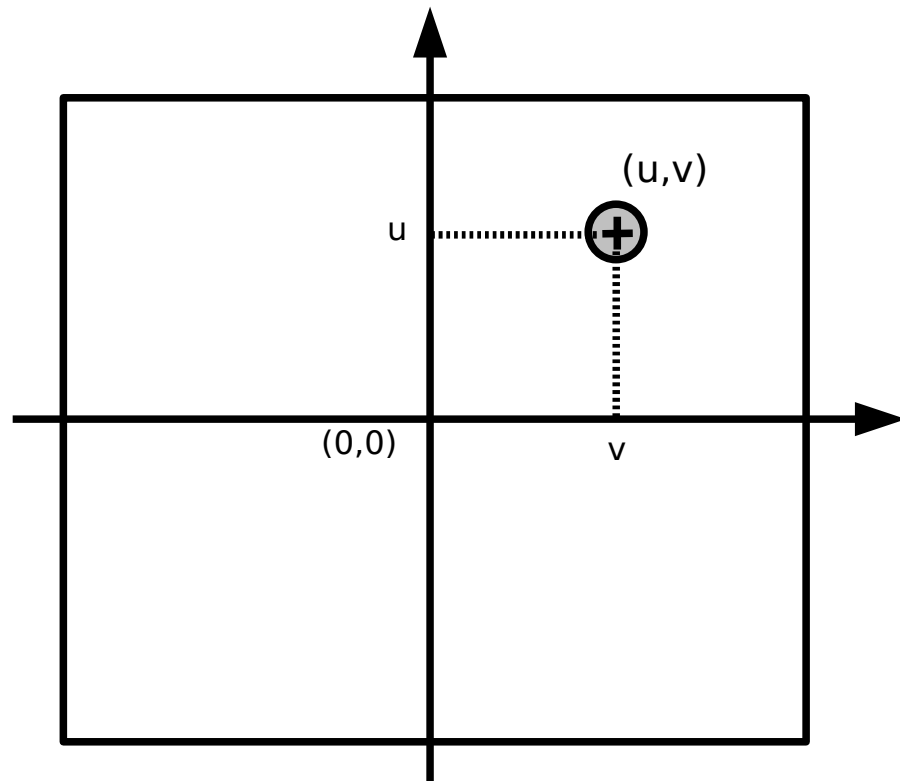


$$u_x = s_\theta$$

$$u_y = s_\phi$$

Control inputs are the attitude references on roll and pitch

# Controller IBVS



Relation between camera motion (6 degree of freedom) and feature motion (on image plane):

$$\begin{bmatrix} \dot{u} \\ \dot{v} \end{bmatrix} = J_V V + J_\omega \Omega$$

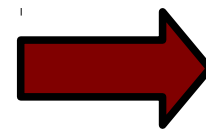
Error on image plane (w.r.t. Image center)

$$e = \begin{bmatrix} e_u \\ e_v \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \end{bmatrix} - \begin{bmatrix} u \\ v \end{bmatrix}$$

Desired camera velocities

$$V_d = J_V^\# \left( -K \begin{bmatrix} u \\ v \end{bmatrix} - J_\omega \Omega \right)$$

$$\dot{e} = -(J_V V_d + J_\omega \Omega) = -K e$$



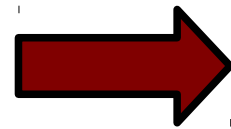
Error exponentially converges to zero

# Controller IBVS

$$u_x = \frac{T}{m} K_a (V_{dx} - V_x)$$

$$u_y = \frac{T}{m} K_a (V_{dy} - V_y)$$

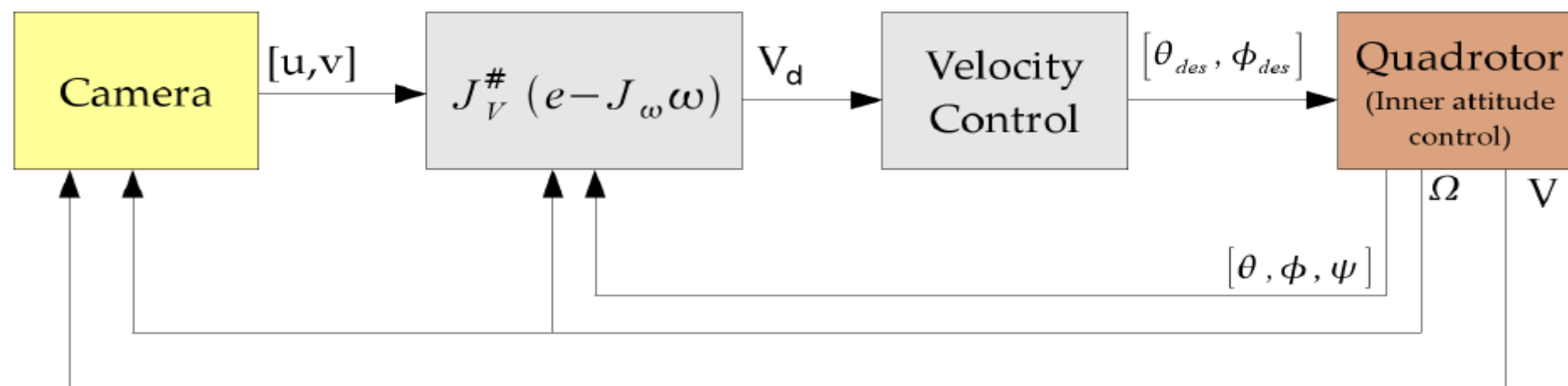
Control inputs



$$\begin{cases} \theta_d = \arcsin \left( \frac{\frac{m}{T} K_a (V_{dx} - V_x)}{\cos \phi} \right) \\ -\phi_d = \arcsin \left( \frac{m}{T} K_a (V_{dy} - V_y) \right) \end{cases}$$

Angular references

## Control scheme

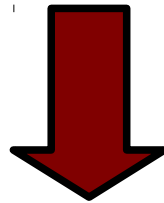




# Main features

- Can be used in unknown environments
- No pose estimation is needed:
  - Linear velocities on the x-y plane are calculated from

$$\hat{V} = J_V^\# \left( \begin{bmatrix} \dot{u} \\ \dot{v} \end{bmatrix} - J_\omega \Omega \right)$$



Low computational cost

Could be implemented on embedded CPU (but requires image analysis)